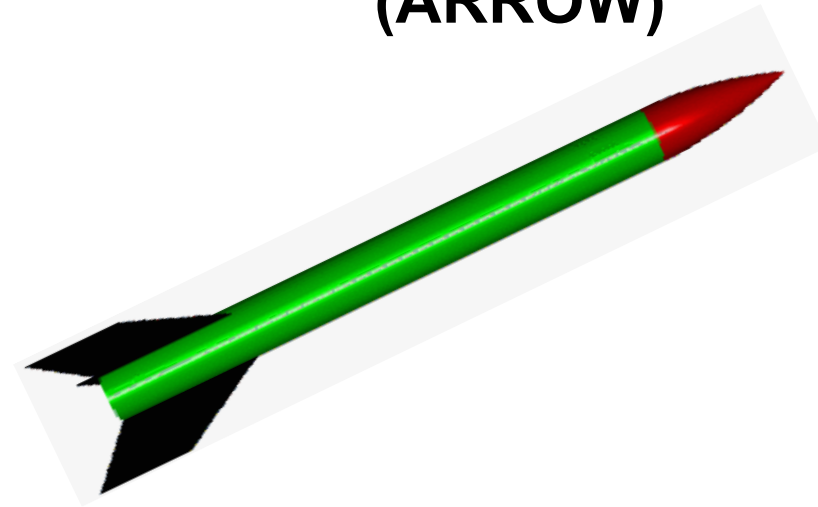
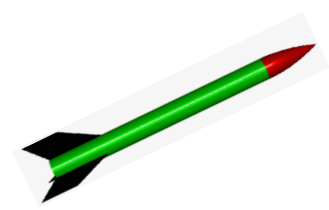




Aerospace Rocket Research for Opportunities in the Workforce Leadership (ARROW)



Base 11 Site Visit
Ms. Ingrid Ellerbe, Senior VP, Partner and Program Engagement
Ms. Tia Tucker, Senior Programs Manager
December 3, 2018

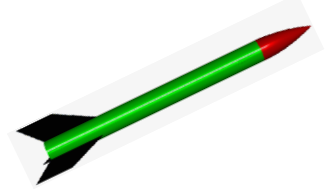


AIM: To Develop Leadership and Workforce in Aerospace Industry

Morgan State University's Strategic Plan:

- 1. Enhancing Student Success:** ARROW will provide unparalleled opportunities for students to engage with innovative academic programs;
- 2. Enhancing Morgan's Status as a Doctoral Research University:** ARROW will provide faculty (as well as students) the opportunity to engage in cutting edge research;
- 3. Improving and Sustaining Morgan's Infrastructure and Operational Processes:** ARROW will provide Morgan an opportunity to address the new need for new program space following the guidelines for environmental sustainability as demanded by the State of Maryland;
- 4. Growing Morgan's Resources:** ARROW will provide additional dynamic opportunities to locate and pursue new grant, contract and entrepreneurial pathways to increase Morgan's resources as well **as enhancing and establishing collaborative relationships with public and private entities**; and
- 5. Engaging with the Community:** ARROW will provide the University an additional tool to be used to reach out, engage and empower the community and the residents that surround Morgan.

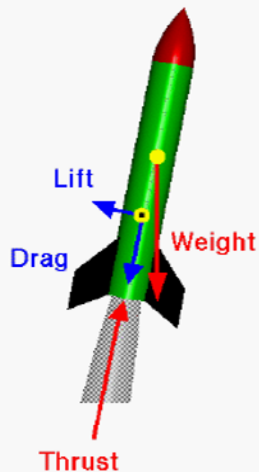
Student Recruitment



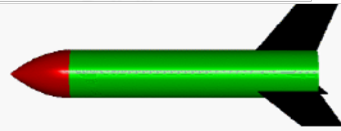
- Morgan State University students with STEM backgrounds
- Partner institution students with STEM backgrounds
- Outreach team

Training

Forces on a Rocket



Newton's Second Law



Differential Form: Force = change of momentum with change of time

$$F = \frac{d(mv)}{dt}$$

or:

Force = change in mass X velocity with time

$$F = \frac{(m_1 V_1 - m_0 V_0)}{(t_1 - t_0)}$$

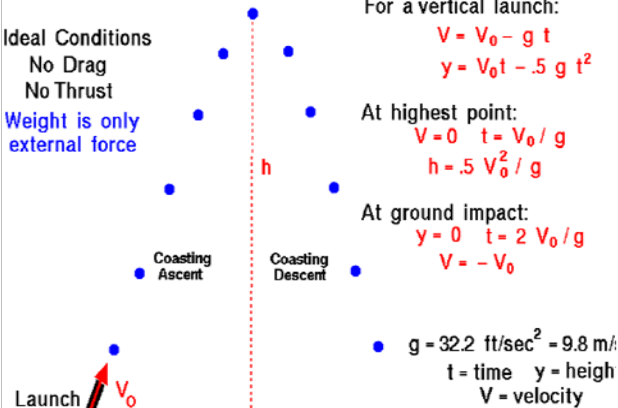
With mass constant: Force = mass X acceleration

$$F = ma$$

Force, acceleration, momentum and velocity are all vector quantities. Each has both a magnitude and a direction.

Ballistic Flight Equations

Ideal Conditions
No Drag
No Thrust
Weight is only external force



For a vertical launch:

$$V = V_0 - g t$$

$$y = V_0 t - .5 g t^2$$

At highest point:

$$V = 0 \quad t = V_0 / g$$

$$h = .5 V_0^2 / g$$

At ground impact:

$$y = 0 \quad t = 2 V_0 / g$$

$$V = -V_0$$

$$g = 32.2 \text{ ft/sec}^2 = 9.8 \text{ m/s}^2$$

t = time y = height
V = velocity

Rocket Thrust Summary

Known: p_t = Total Pressure R = Gas Constant
 T_t = Total Temperature A = Area
 γ = Specific Heat Ratio

$$\text{Mass Flow Rate: } \dot{m} = \frac{A^* p_t}{\sqrt{T_t}} \sqrt{\frac{\gamma}{R}} \left(\frac{\gamma+1}{2}\right)^{-\frac{\gamma+1}{2(\gamma-1)}}$$

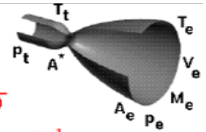
$$\text{Exit Mach: } \frac{A_e}{A^*} = \left(\frac{\gamma+1}{2}\right)^{-\frac{\gamma+1}{2(\gamma-1)}} \frac{(1 + \frac{\gamma-1}{2} M_e^2)^{\frac{\gamma+1}{2}}}{M_e}$$

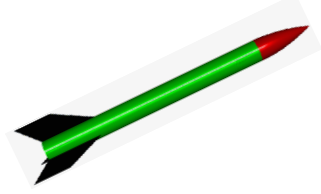
$$\text{Exit Temperature: } \frac{T_e}{T_t} = \left(1 + \frac{\gamma-1}{2} M_e^2\right)^{-1}$$

$$\text{Exit Pressure: } \frac{p_e}{p_t} = \left(1 + \frac{\gamma-1}{2} M_e^2\right)^{-\frac{\gamma}{\gamma-1}}$$

$$\text{Exit Velocity: } V_e = M_e \sqrt{\gamma R T_e}$$

$$\text{Thrust: } F = \dot{m} V_e + (p_e - p_o) A_e$$





| Altitude Phases | STAGES OF ROCKET BUILD | | |
|------------------------------|---|---|---|
| | 13,000 | 100,000 | 150,000 |
| Phase I: Design | Lead: Dr. Damoah Co-Lead: Dr. Willoughby Students (n = 4) | Lead: Dr. Damoah Co-Lead: Dr. Willoughby Students (n = 4) | Lead: Dr. Damoah Co-Lead: Dr. Willoughby Students (n = 4) |
| Phase II: Build | Lead: Dr. Willoughby Co-Lead: Dr. Damoah Students (n = 4) | Lead: Dr. Willoughby Co-Lead: Dr. Damoah Students (n = 4) | Lead: Dr. Willoughby Co-Lead: Dr. Damoah Students (n = 4) |
| Phase III: Safety | Lead: Dr. Lee Co-Lead: Dr. Chen Students (n = 4) | Lead: Dr. Lee Co-Lead: Dr. Chen Students (n = 4) | Lead: Dr. Lee Co-Lead: Dr. Chen Students (n = 4) |
| Phase IV: Launch | Lead: Dr. Willoughby Co-Lead: Dr. Kinyua Students (n = 4) | Lead: Dr. Willoughby Co-Lead: Dr. Kinyua Students (n = 4) | Lead: Dr. Willoughby Co-Lead: Dr. Kinyua Students (n = 4) |

Phase I: Design



SEB 104 Design Stations



3D printer

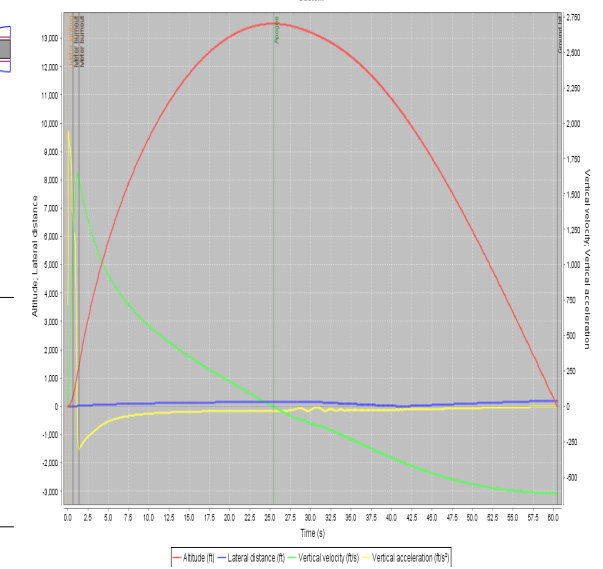
Rocket Design



Rocket
Stages: 1
Mass (with motor): 7048 g
Stability: 2.98 cal
CG: 96.2 cm
CP: 123 cm

| | Motor | Avg Thrust | Burn Time | Max Thrust | Total Impulse | Thrust to Wt | Propellant Wt | Size |
|------------------------|----------|------------|-----------|------------|---------------|--------------|---------------|-----------|
| Altitude | 4120 m | | | | | | | |
| Flight Time | 60.5 s | H550 | 553 N | 0.567 s | 643 N | 313 Ns | 7.99:1 | 176 g |
| Time to Apogee | 25.3 s | I435T | 470 N | 1.27 s | 702 N | 596 Ns | 6.80:1 | 287 g |
| Optimum Delay | 24 s | | | | | | | 38/298 mm |
| Velocity off Pad | 35.1 m/s | J605F | 586 N | 1.26 s | 896 N | 738 Ns | 8.48:1 | 367 g |
| Max Velocity | 501 m/s | | | | | | | 38/476 mm |
| Velocity at Deployment | N/A | 1500 | 508 N | 0.683 s | 1155 N | 338 Ns | 7.35:1 | 748 g |
| Landing Velocity | 187 m/s | J1026 | 1039 N | 1.21 s | 1193 N | 1264 Ns | 15.03:1 | 616 g |
| | | | | | | | | 38/625 mm |
| Total: | | | | | 3249 Ns | 45.64:1 | 2194 g | |

Simulation 8



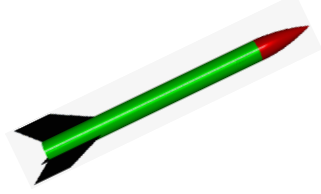
Earthquake simulator



Wind Tunnel

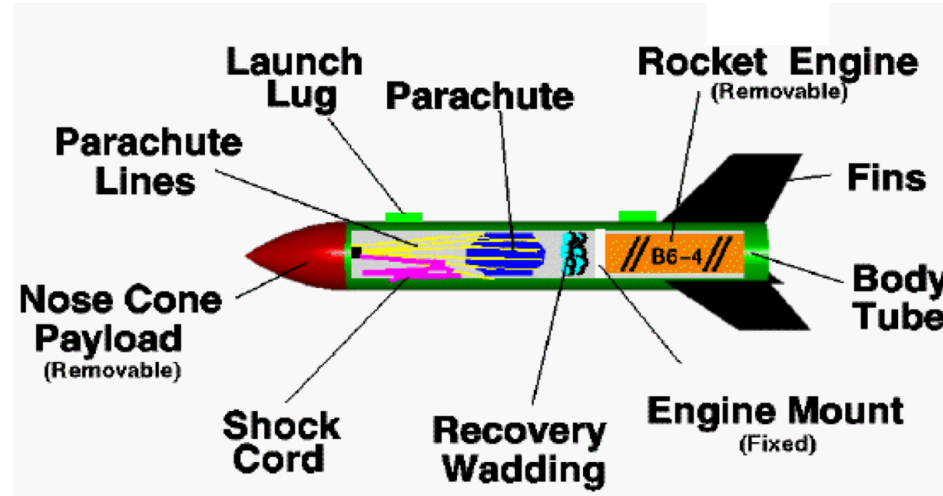
- CAD/CAM (Drawing Generation)
- CFD (Theory Analysis)
- 3D Printer (Prototype)
- Wind Tunnel (Experimental Analysis)
- Rocket design simulator (like OpenRocket or ANSYS)

Phase II: Build



Rocket Engine Components

- Air Frame
- Payload
- Recovery System

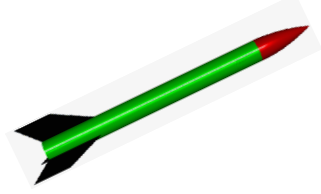


CBEIS 018: Component Fabrication



MEB 136 Lab: Components Assembly

Phase II: Build



Arc-Welding Machine



Injection Molding Machine

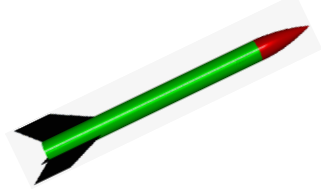


CNC Cutting Machine



Other Tools

Phase III: Safety



Establish Morgan Safety Council

- Vice President of Research and Economic Development
- Manager of Building Construction and Management
- Fire Marshall
- Lead faculty
- Supporting staff

Conduct Safety Training

Observe Safety Precautions

Contact Information for Safety Related Issues

Fire Marshal

Christopher G. Evans

Fire Life and Environmental Health Safety

Programs - DCM

Phone: 443-885-4451

Cell: 443-757-7486, Fax: 443-885-8269

University Police

Phone: 443-885-3103

For Minor Accidents

University Health Center

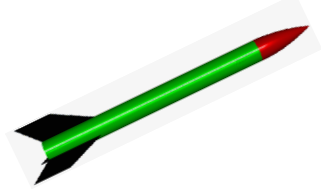
For Serious Accidents Dial:

911 or 0

Hospital & Ambulance

Med-Star Good Samaritan Hospital

Phase III: Safety



Create safety guidelines and procedures

- General rules
- Work areas and apparatus
- Injuries and & accident guidelines
- Emergency procedures
- Personal protective equipment
- Guidelines to use tools and equipment



Fire Extinguisher



Gas Masks Protection



Eye Protection Goggles



Dust Protection Masks



First Aid Kit Box



High Temperature Gloves



General Machine Gloves

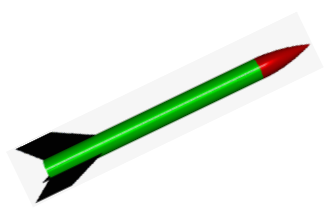


Head Protection



Hearing Protection

Phase IV: Launch



1. Training and Combustion Testing
2. Launch Location Coordination
 - Goddard Space Flight Center or
 - Wallops Space Flight Center
3. Launch Window Determination
 - ❖ Risk Analysis
 - ❖ Determining **go or no-go flight launch** at
 - **our flight headquarters.**
4. Recovery by Streamer, Parachute or GPS
5. Evaluation